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# Development of A Transient Analysis Code for S-CO<sub>2</sub> Power Conversion System

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- ◆ Introduction
- ◆ SCTRAN/CO<sub>2</sub> development
- ◆ Initial verification for component model
- ◆ Initial verification for loop simulation
- ◆ Conclusion & expectation

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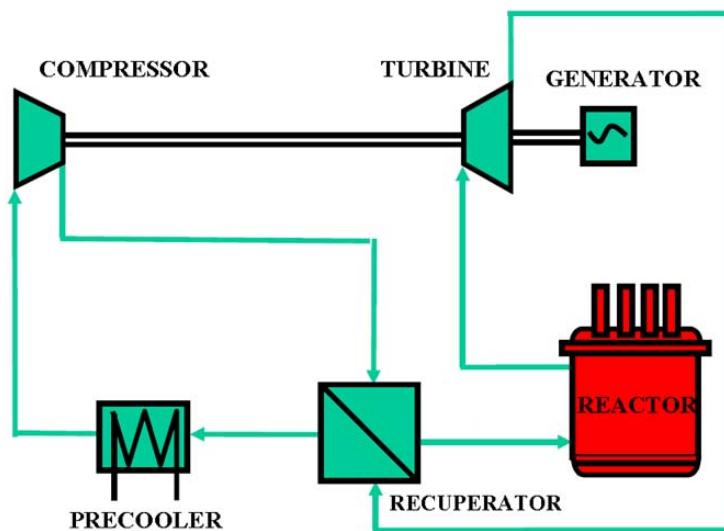
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# PART 1 INTRODUCTION

## ■ S-CO<sub>2</sub> Brayton Cycle

### S-CO<sub>2</sub> Brayton Cycle Advantage:

- ✓ High thermal efficiency
- ✓ Simple configuration
- ✓ Compact turbomachinery



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*Transient analysis code used in S-CO<sub>2</sub> Brayton Cycle*

| Built up method                                 | Analysis code             | Applied in  |
|---|---------------------------|---|
| Developed with an exist Transient analysis code | TRACE                     | S-CO <sub>2</sub> Brayton cycle                                     |
|   | GAMMA+                    | KAIST Micro Modular Reactor(MMR)                                    |
|   | MARS                      | Supercritical CO <sub>2</sub> Integral Experimental Loop (SCIEL)    |
|   | RELAP5-3D                 | SCO <sub>2</sub> cooled fast reactors                               |
|   | MMS-LMR                   | Sodium cooled fast reactor KALIMER-600                              |
|   | GAS-PASS                  |   |
| Developed with nothing                          | Plant Dynamics Code (PDC) | S-CO <sub>2</sub> Brayton cycle coupled to lead-cooled fast reactor |



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# PART 2 SCTRAN/CO<sub>2</sub> Development

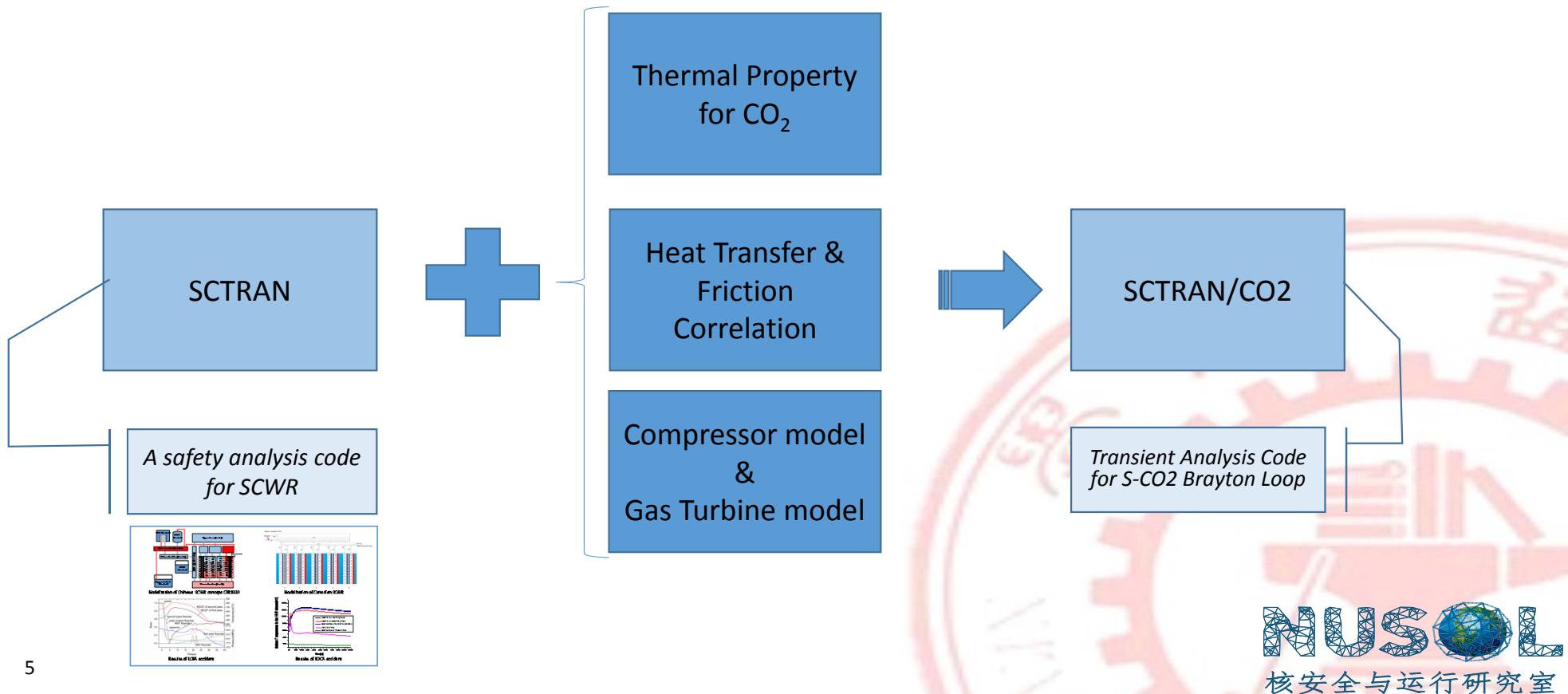
- SCTRAN introduction
- Component model needed for SCTRAN/CO<sub>2</sub>
  - ✓ Constitutive model
  - ✓ Compressor model
  - ✓ Gas turbine model
  - ✓ Shaft model

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# 2.1 SCTRAN Introduction



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## 2.2 Constitutive Model

### □ Property of carbon dioxide

$$F(p, h) = \sum_{i=0}^4 \sum_{j=0}^4 a_{ij} p^i h^j, \quad p \leq p_{critical}, h < h_i(p) \text{ 或 } p > p_{critical}, h \leq h_{set10}$$

$$F(p, h) = \sum_{i=0}^4 \sum_{j=0}^4 b_{ij} p^i h^j, \quad p \leq p_{critical}, h_g(p) < h \leq h_{set20}$$

或  $p > p_{critical}, h_{set11} < h \leq h_{set20}$

$$F(p, h) = \sum_{i=0}^4 \sum_{j=0}^4 c_{ij} p^i h^j, \quad h > h_{set21}$$

$$F(p, h) = F\left(\rho, T\right) \quad F = \sum_{i=0}^4 \sum_{j=0}^4 a_{ij} T^i \rho^j \quad - F(p, h_{set10})]$$

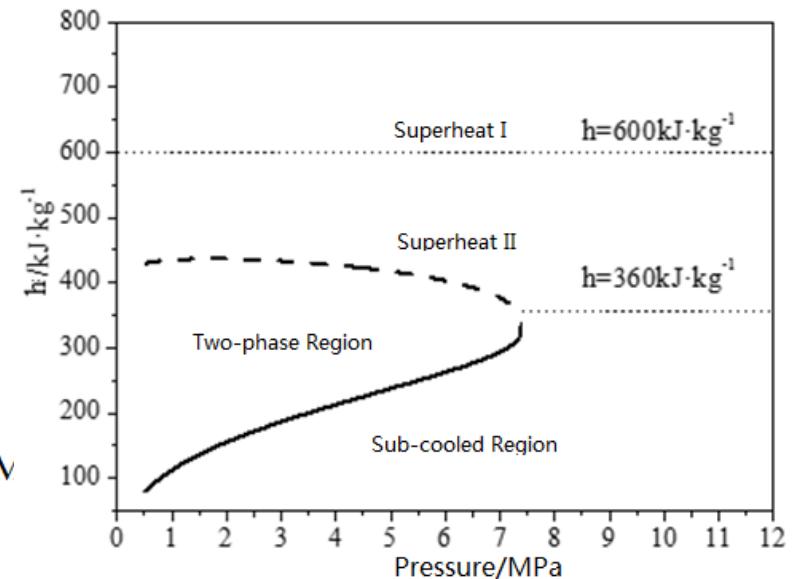
$\rightarrow p > p_{critical} > h_{set10} > h_{set11}$

$$F(p, h) = F(p, h_{set20}) + \frac{(h - h_{set20})}{(h_{set21} - h_{set20})} [F(p, h_{set21}) - F(p, h_{set20})]$$

$\rightarrow h_{set20} < h \leq h_{set21}$

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$$5 < p < 5.5 \text{ MPa}$$



## 2.2 Constitutive Model

### □ Heat transfer correlation

Gnielinski Correlation:

$$N_u = \frac{hD_e}{\lambda} = \frac{(f/8)(Re-1000)\Pr}{1+12.7\sqrt{(f/8)(\Pr^{2/3}-1)}} , 2300 < Re < 5 \times 10^6 , 0.5 < \Pr < 200$$

### □ Friction correlation

$$\frac{1}{\sqrt{f}} = -2 \log \left\{ \frac{\varepsilon}{3.7D_e} + \frac{2.51}{Re} \left[ 1.14 - 2 \log \left( \frac{\varepsilon}{D_e} + \frac{21.25}{Re^{0.9}} \right) \right] \right\} , Re > 3400$$

$$f = \frac{64}{Re} , Re < 2300$$

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## 2.3 Compressor Model

- ❑ Compressor model : Solution
- ✓ Compressor torque

$$\tau_t = \tau_s + \tau_d = \frac{m}{\omega} (h_{2s} - h_{01}) + \frac{m}{\omega} (h_{02} - h_{2s})$$

- ✓ Ideal outlet fluid enthalpy

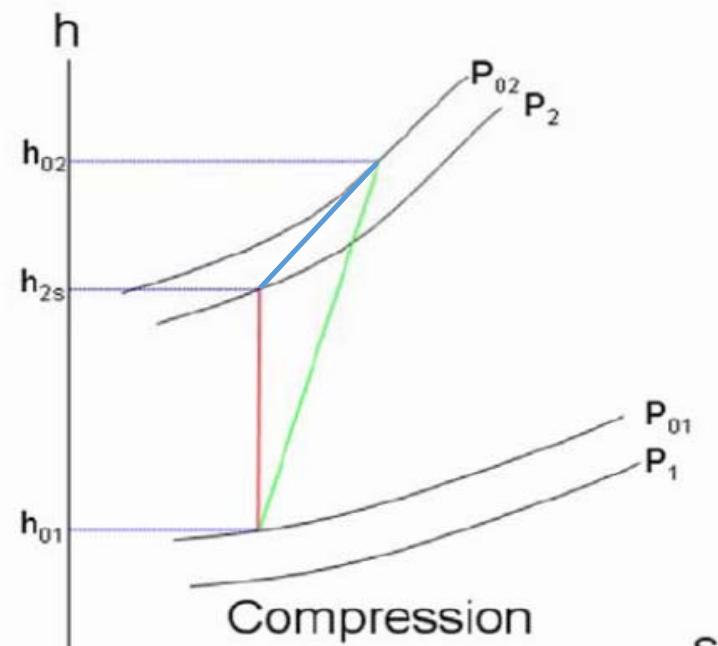
$$h_{2s} = h_{01} + \int_{P_1}^{P_2} v_m * dp$$

among,  $P_2 = P_1 * R_p$

- ✓ Realistic outlet fluid enthalpy

$$\eta_{ad} = \frac{h_{2s} - h_{01}}{h_{02} - h_{01}}$$

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*Ideal and realistic compression process  
inside compressor*



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## 2.3 Compressor Model

### □ Compressor model : Intergrated in SCTRAN

- ✓ Total torque of compressor

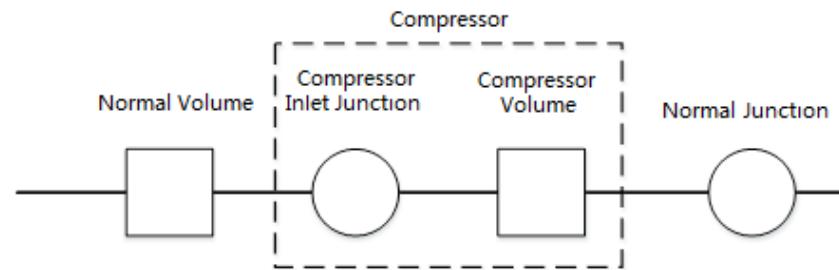
$$\tau_t = \tau_s + \tau_d = \frac{\dot{m}}{\omega} (h_{2s} - h_{01}) + \frac{\dot{m}}{\omega} (h_{02} - h_{2s}) = \frac{\dot{m}}{\omega \eta_{ad}} (h_{2s} - h_{01}) = \frac{\dot{m}}{\omega \eta_{ad}} \frac{P_1^T (R_p - 1)}{\rho_m} = \frac{\dot{m}}{\omega \eta_{ad}} \frac{(P_2 - P_1)}{\rho_m}$$

- ✓ Compressor work added on fluid

$$W = \tau_t * \omega$$

- ✓ Pressure rise

$$\Delta P = P_1 (R_p - 1)$$



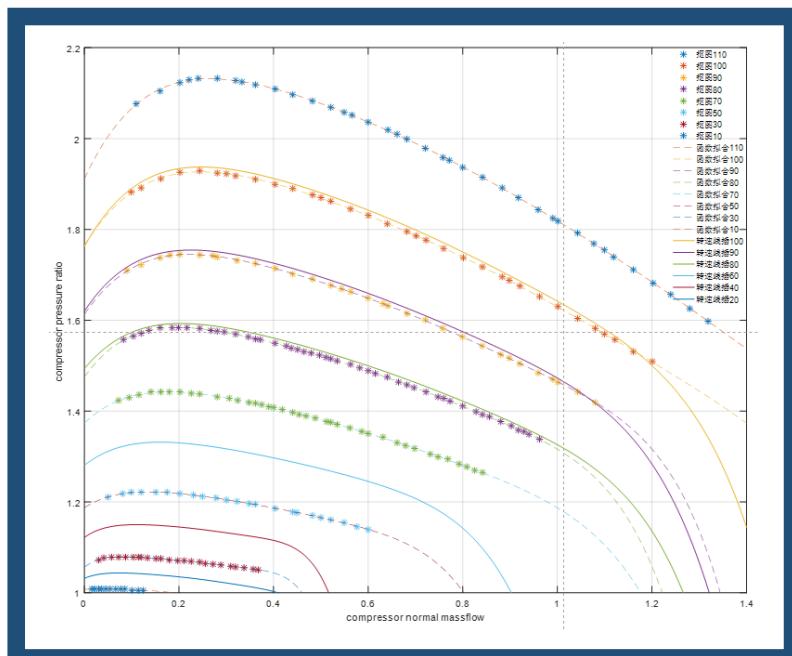
*The compressor incorporated in SCTRAN/CO2*



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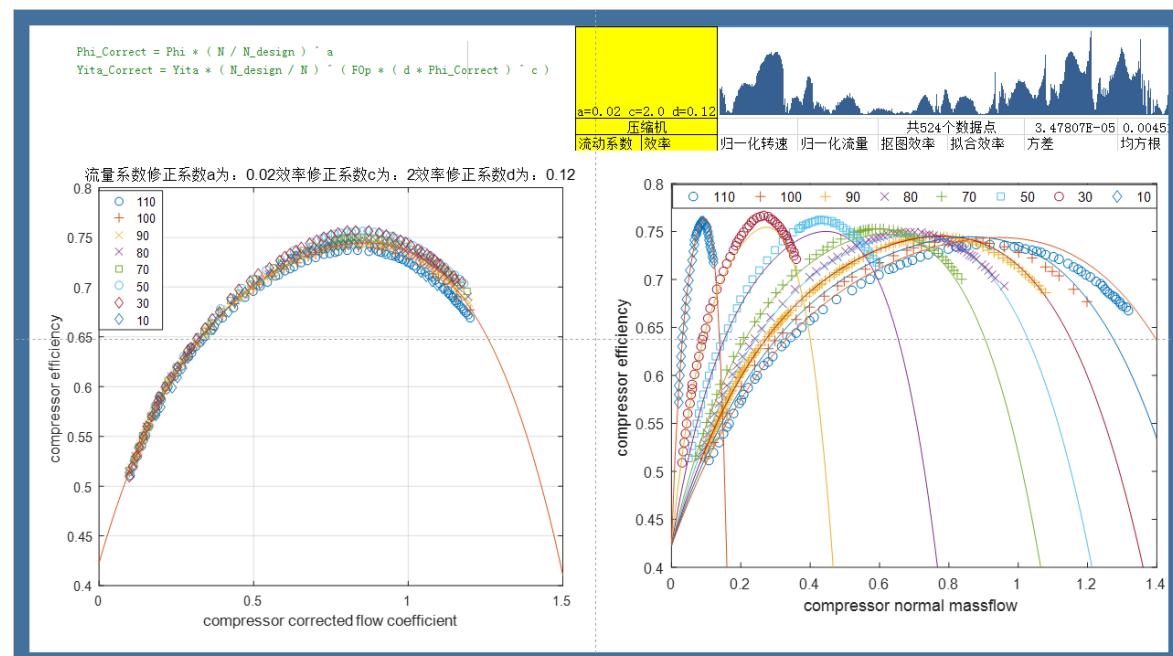
## 2.3 Compressor Model

### □ Compressor model : Performance Map



Compressor Pressure Ratio

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Compressor Efficiency



## 2.4 Turbine Model

- ✓ fluid enthalpy increase

$$\Delta h = \frac{h_2^T - h_1^T}{\eta_{ad}}$$

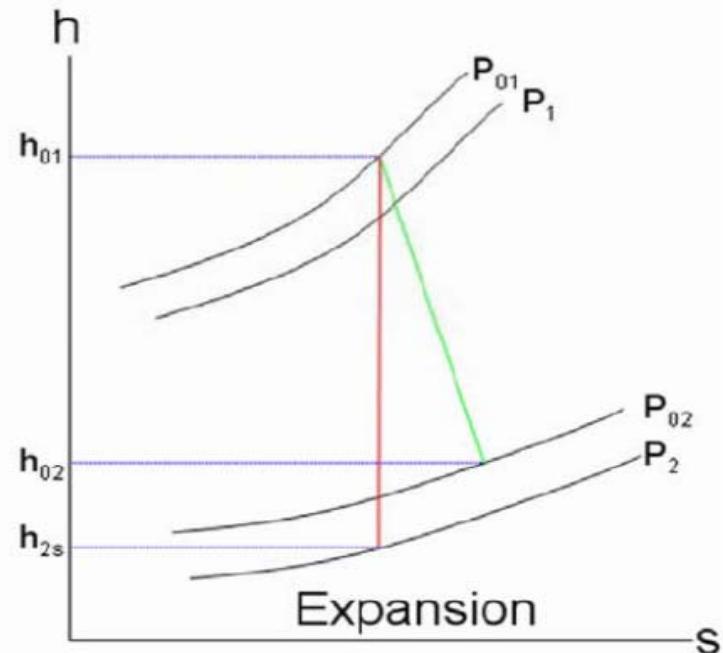
- ✓ pressure drop

$$\Delta P = P_1 (R_p - 1)$$

- ✓ total torque of gas turbine

$$\tau = \frac{m \eta (P_1 - P_2)}{\omega \rho_m}$$

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*Ideal and realistic expansion process  
inside gas turbine*

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## 2.5 Shaft Model

- ✓ Mode 1(without control system)

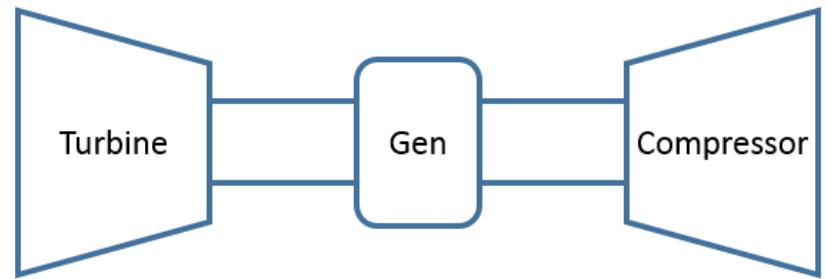
$$\omega_{T,i} = \omega_{C,k} = \omega_{Shaft} = \text{User defined}$$

- ✓ Mode 2(with control system)

$$\sum_i I_I \frac{d\omega}{dt} = \sum_m \tau_{T,m} - \sum_n \tau_{C,n} + \tau_g$$

Among:

$$\tau_g = C * \tau_{g,i}$$



# PART 3 COMPONENT MODEL VERIFICATION

- Thermal property verification
- PCHE model verification
- Compressor model verification

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# 3.1 Thermal Property Package Verification

*Relative prediction error of the developed CO<sub>2</sub> property package compared to NIST REFPROP 9.0*

| CO <sub>2</sub> Property  | Symbol    | Regions              | Relative Error   |
|---------------------------|-----------|----------------------|--|
| Saturated Liquid Enthalpy | $h_f$     | -                    | ±0.015%  |
| Saturated Vapor Enthalpy  | $h_g$     | -                    | ±0.009%  |
| Temperature               | T         | Subcooled area       | (-0.05%, 0.1%), 99% of which is within relative errors of ±0.05%   |
|                           |           | Superheated region 1 | (-0.2%, +0.2%) , 99% of which is within relative errors of ±0.1%   |
|                           |           | Superheated region 2 | (-0.1%, 0.25%), 99% of which is within relative errors of ±0.05%   |
| Specific Volume           | v         | Subcooled area       | (-0.5%, 1%) , 99% of which is within relative errors of ±0.5%      |
|                           |           | Superheated region 1 | (-1%, 4%) , 99% of which is within relative errors of ±1.0%        |
|                           |           | Superheated region 2 | (-0.5%, 0.1%) , 99% of which is within relative errors of ±0.1%    |
| Thermal Conductivity      | $\lambda$ | -                    | (-30%, 40%) near the critical region , (-2%, +2%) at other regions |
| Dynamic Viscosity         | $\mu$     | -                    | (-1.5%, 0.5%) , 99% of which is within relative errors of ±0.5%    |

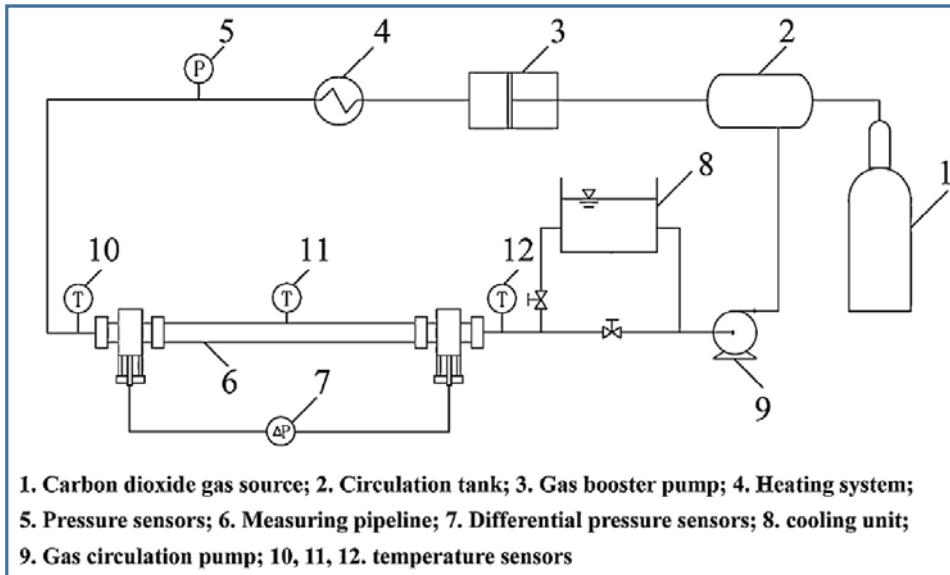


## 3.2 PCHE Model Verification

### ■ Friction model code programming verification

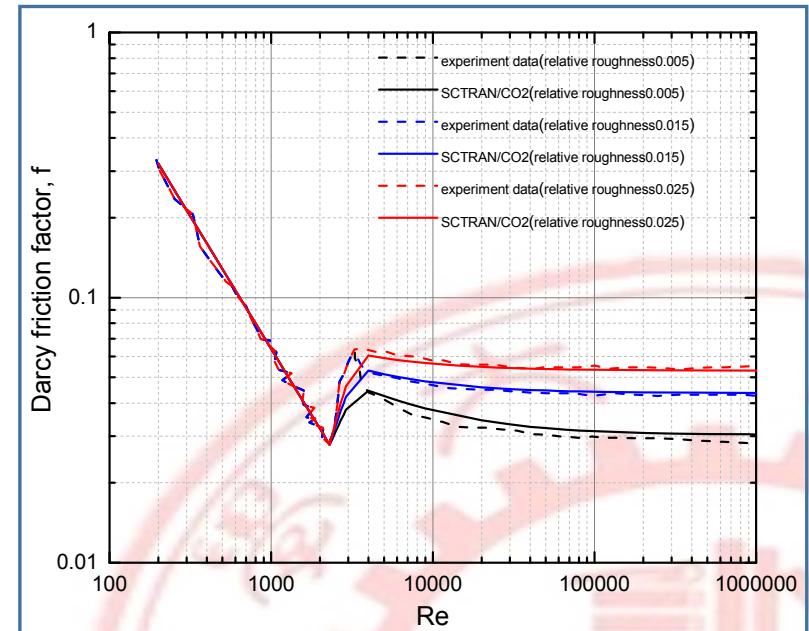
#### Experimental Conditions:

The temperature range : 30-150°C;  
The pressure range : 3.5-40 MPa;  
The Reynolds number range :  $200-2.0 \times 10^6$ ;  
The surface relative roughness (ratio of roughness over tube diameter) : 0.005, 0.015 and 0.025.



Wang et al. Experimental Loop

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Comparison with experimental data  
for friction coefficient of various roughness



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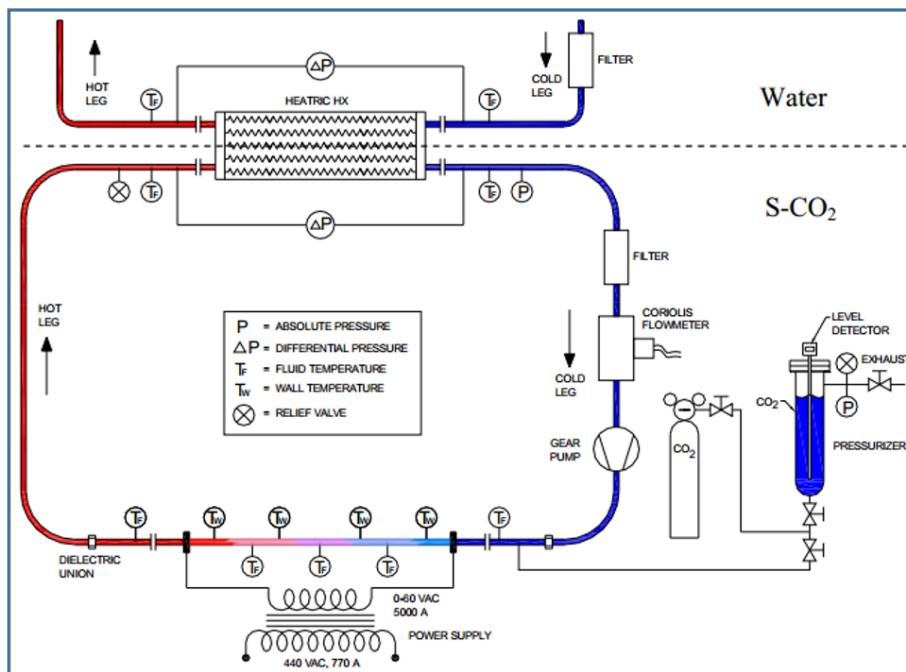
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## 3.2 PCHE Model Verification

### ■ PCHE model verification



*JOSH VAN METER PCHE Experimental Loop*

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**Built in:**

ANL

**Composed of:**

Cooling water system

$\text{CO}_2$  Circle System

Pressure Stabilizing System

**Focused on:**

Water and  $\text{CO}_2$  heat transfer  
characteristic in PCHE



## 3.2 PCHE Model Verification

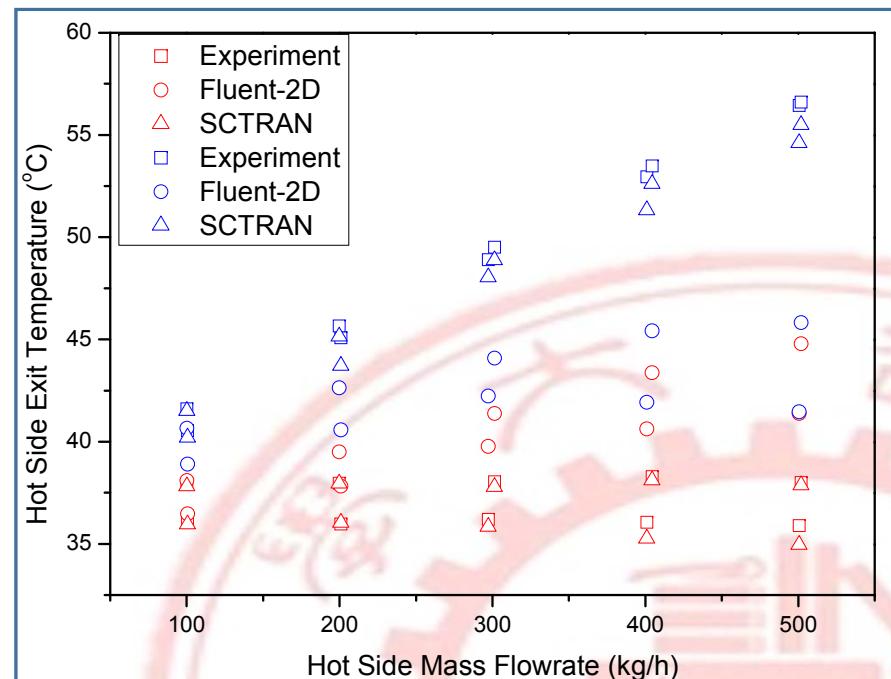
### ■ PCHE model verification

| TEST_NO. | CO <sub>2</sub> Side |          |         | H <sub>2</sub> O Side |         |
|----------|----------------------|----------|---------|-----------------------|---------|
|          | Pressure             | Flowrate | Temp_In | Flowrate              | Temp_In |
| B6       | 8.003                | 100.53   | 88.63   | 701.59                | 35.63   |
| B7       | 8.001                | 200.77   | 88.1    | 699.78                | 35.11   |
| B8       | 7.972                | 297.14   | 89.36   | 701.8                 | 35.05   |
| B9       | 8.003                | 401.01   | 87.92   | 701.77                | 33.28   |
| B10      | 7.995                | 500.61   | 87.93   | 700.09                | 31.28   |
| B11      | 8.003                | 100.03   | 87.68   | 697.8                 | 37.68   |
| B12      | 8.005                | 199.73   | 88.85   | 697.8                 | 37.53   |
| B13      | 7.998                | 301.31   | 88.17   | 699.86                | 37.48   |
| B14      | 8.02                 | 404.29   | 88.97   | 701.62                | 37.58   |
| B15      | 7.998                | 501.79   | 88.09   | 702.25                | 36.83   |

*Test Conditions :*

*PCHE fluid inlet temperatures and Mass flowrate*

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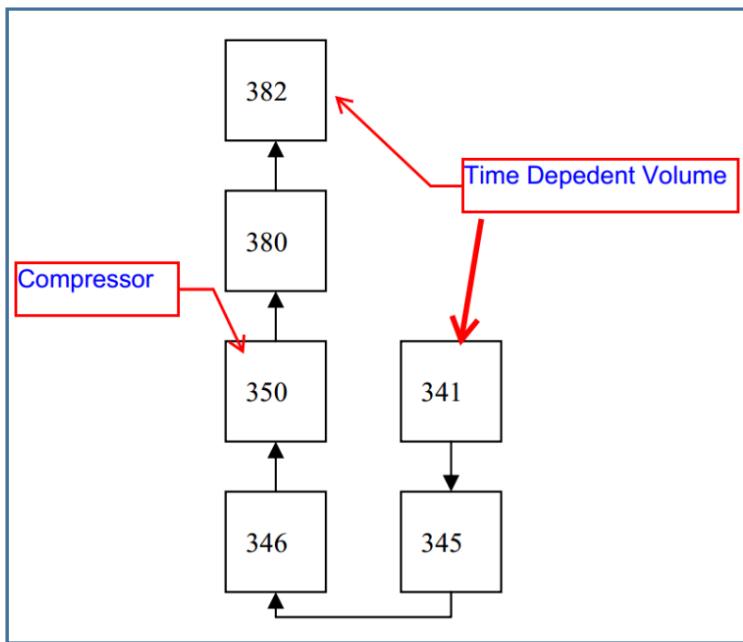


*PCHE modeling results : PCHE fluid outlet temperatures*



# 3.3 Compressor Model Verification

## ■ Compressor model verification



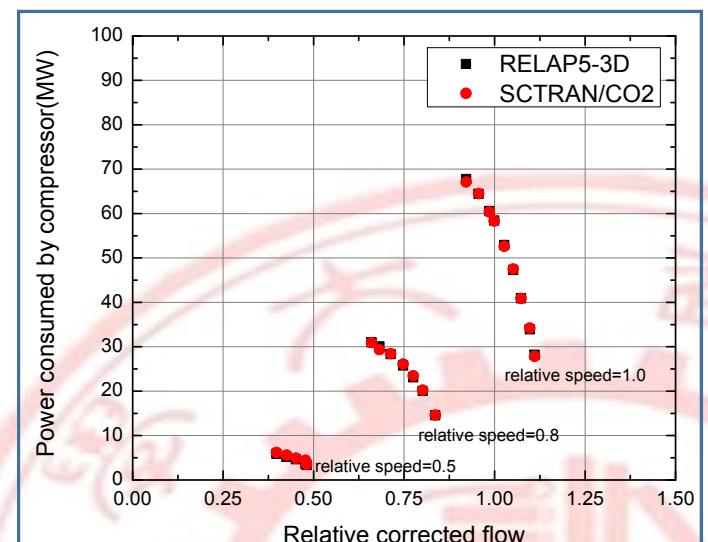
18 *Nodalization of  
the recompressing compressor*

**Boundary conditions:**  
TDV 341: 9.08MPa, 363K

**Operation parameters:**  
Relative flowrate: 0.4-1.0  
Relative speed: 0.5, 0.8, 1.0

**Result:**  
The compressor model in SCTRAN/CO2 is able to predict the compressor consuming power.

$$W_{c,v} = \dot{m} \frac{1}{\eta_{ad}} \frac{P_1 (R_p - 1)}{\rho_m}$$



*Predicted compressor consuming power  
by SCTRAN/CO2 and RELAP5-3D*

# PART 4 LOOP SIMULATION VERIFICATION

➤ S-CO<sub>2</sub> PE Loop

➤ IST Loop

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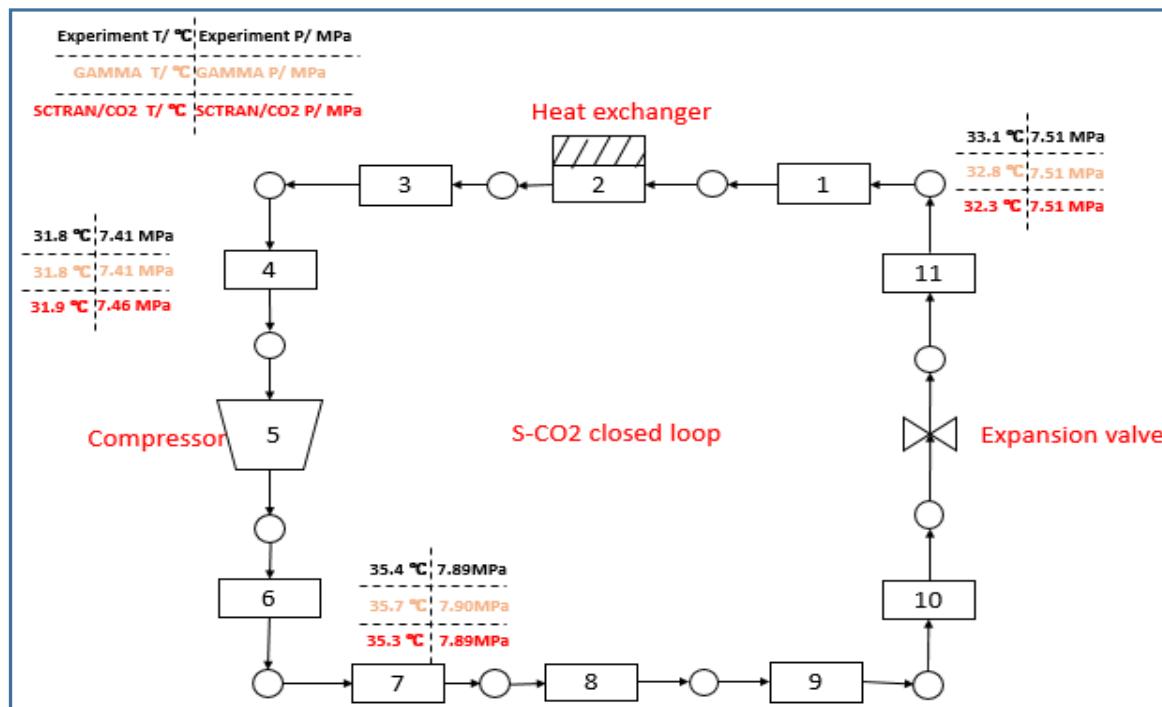
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# 4.1 S-CO<sub>2</sub> PE Loop Simulation Verification

## ■ S-CO<sub>2</sub> PE Loop Steady State Simulation



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*SCO<sub>2</sub>PE loop simulation nodalization and steady result*

### Simplification:

1. SCTRAN/CO<sub>2</sub> applies a heat flux boundary to simulate the heat exchanger in the steady
2. The pressure ratio and efficiency keeps constant in the steady and transient simulation

### Steady Result:

1. The Temp Error is within 0.2 °C
2. The Pressure Error is within 0.1 MPa



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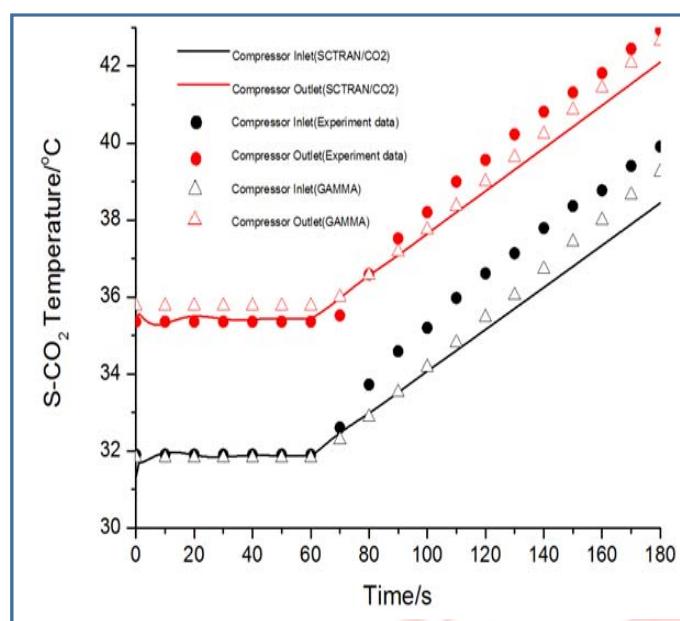
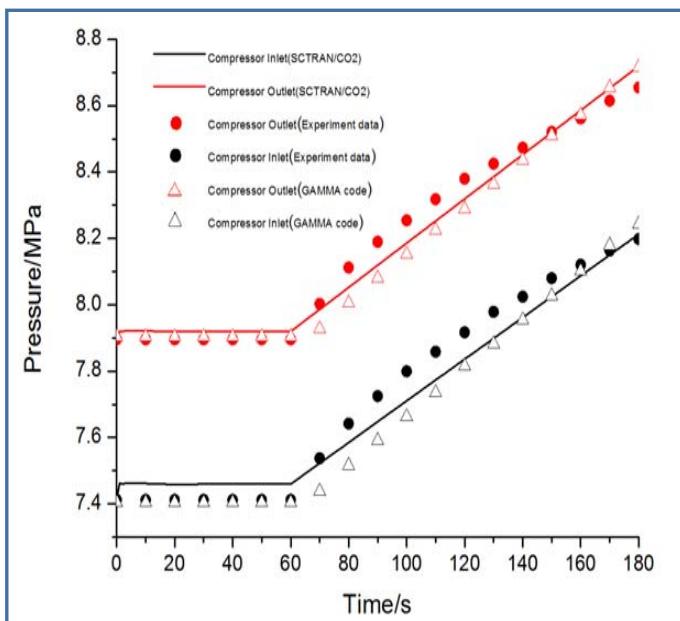
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# 4.1 S-CO<sub>2</sub> PE Loop Simulation Verification

## ■ S-CO<sub>2</sub> PE Loop Transient Simulation



*Pressure and temperature variation during the cooling reduction transient*

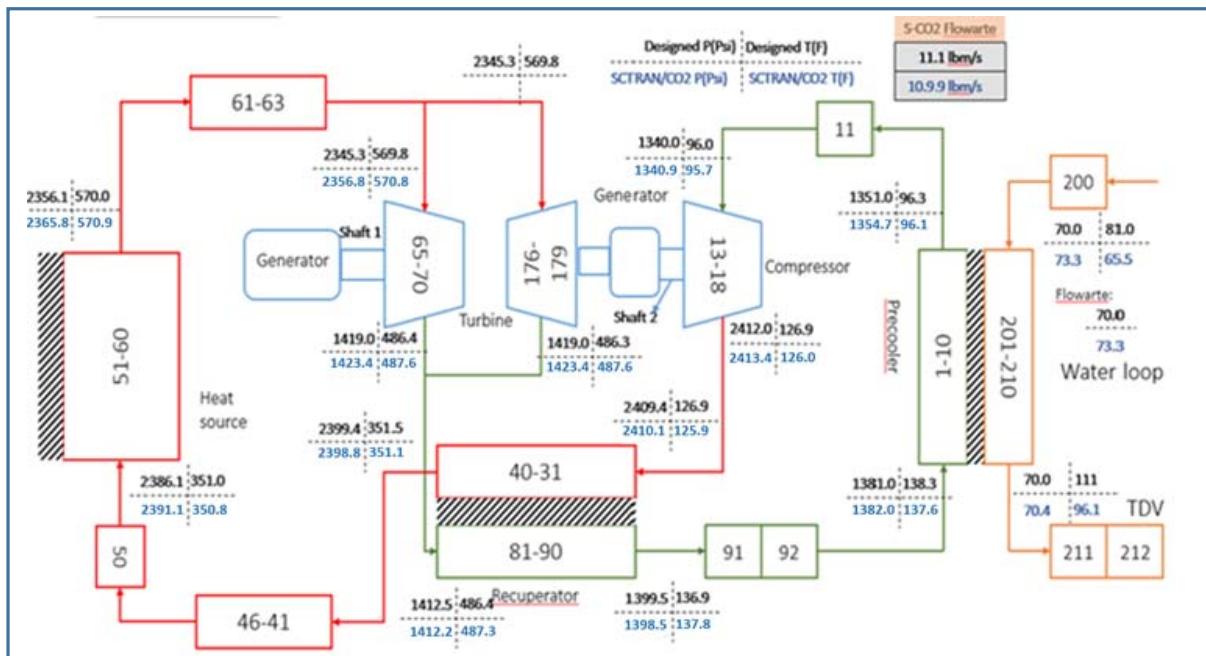
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**Transient:**  
water flowrate from  
0.25 kg/s to 0.17 kg/s in  
60 second

**Result:**  
the relative error of  
pressure is within 1% ;  
the error of temp is  
within 2 °C.

# 4.2 IST Loop Simulation Verification

## ■ IST Loop Full Power Heat Balance Simulation



| Designed                             | SCTRAN/CO <sub>2</sub> | Error  |
|--------------------------------------|------------------------|--------|
| CO <sub>2</sub> Loop Flowrate(lbm/s) |                        |        |
| 11.1                                 | 10.99                  | -0.99% |
| Max Temperature Difference(F)        |                        |        |
| 486.4                                | 487.6                  | 1.2    |
| Max Pressure Difference(psi)         |                        |        |
| 2345.3                               | 2356.8                 | 1.2%   |

### Conclusion:

1. The SCTRAN/CO<sub>2</sub> is able to simulate S-CO<sub>2</sub> Brayton cycle
2. Transient process isn't presented



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# PART 5 CONCLUSION & EXPECTATION

## ■ Conclusion

- ✓ The PCHE model can predict the fluid outlet temperature at steady state.
- ✓ The compressor model of SCTRAN/CO<sub>2</sub> can predict accurate compressor consuming power, which indicate it can be used for Brayton cycle simulation.
- ✓ Transient simulation of SCO<sub>2</sub>PE and steady state simulation of IST indicate that SCTRAN/CO<sub>2</sub> owns the ability to conduct transient simulations for S-CO<sub>2</sub> Brayton cycle.

## ■ Expectation

- ✓ The friction model for PCHE model should be validated
- ✓ The transient validation for PCHE model is wanted
- ✓ To do some control strategy analysis for brayton cycle with our newly developed code

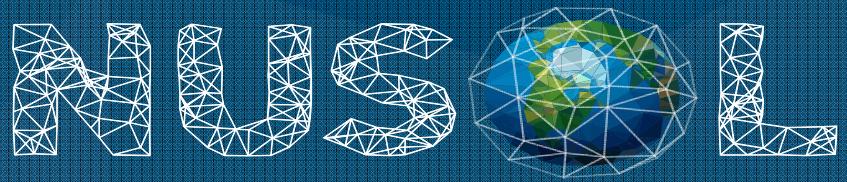
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# THANK YOU

Welcome your suggestions!



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